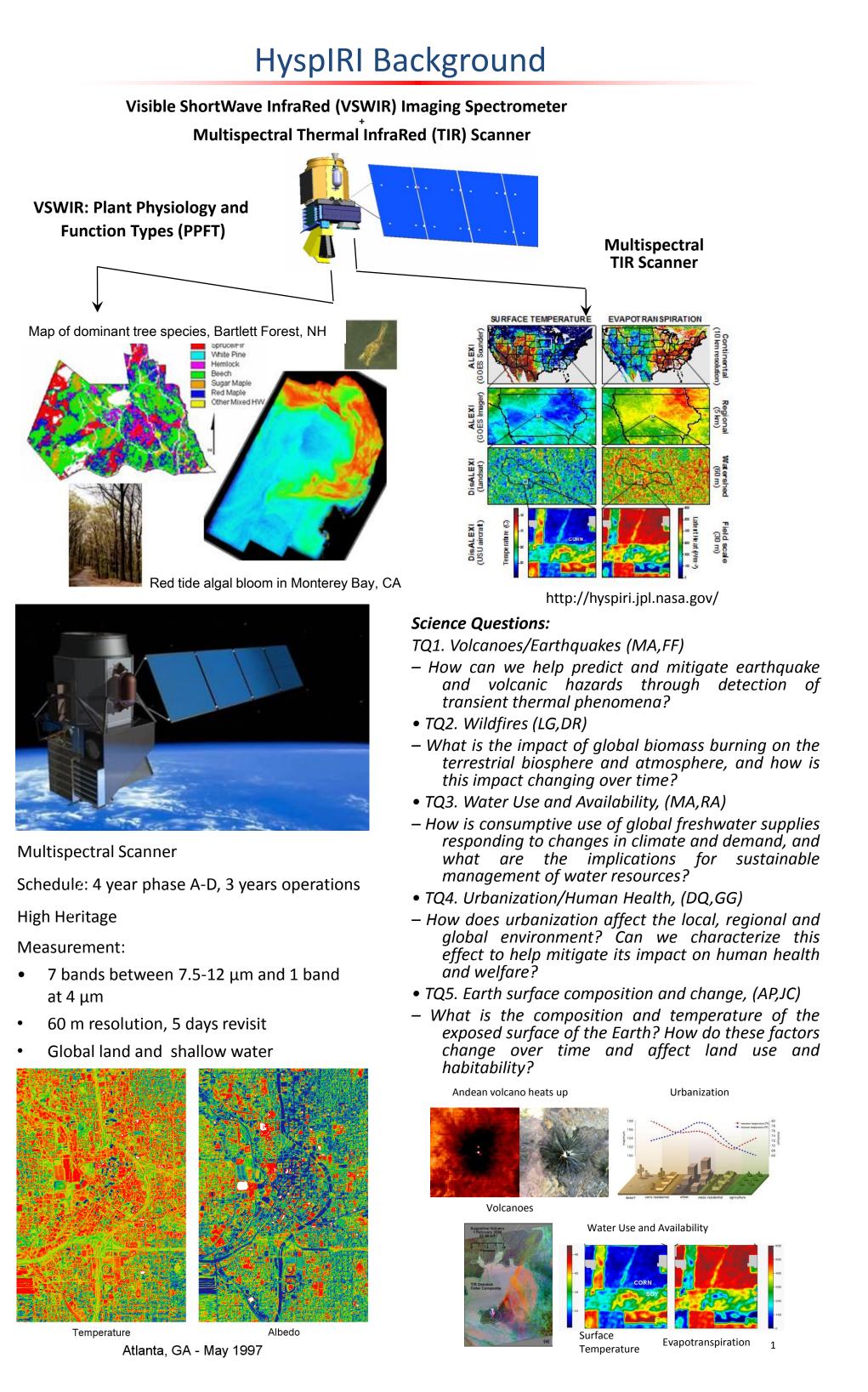


The Prototype HyspIRI Thermal Infrared Radiometer (PHyTIR)

Simon J. Hook, Bjorn T. Eng, Bruno Jau, Marc Foote & William R. Johnson



PHyTIR Rational for HyspIRI-TIR

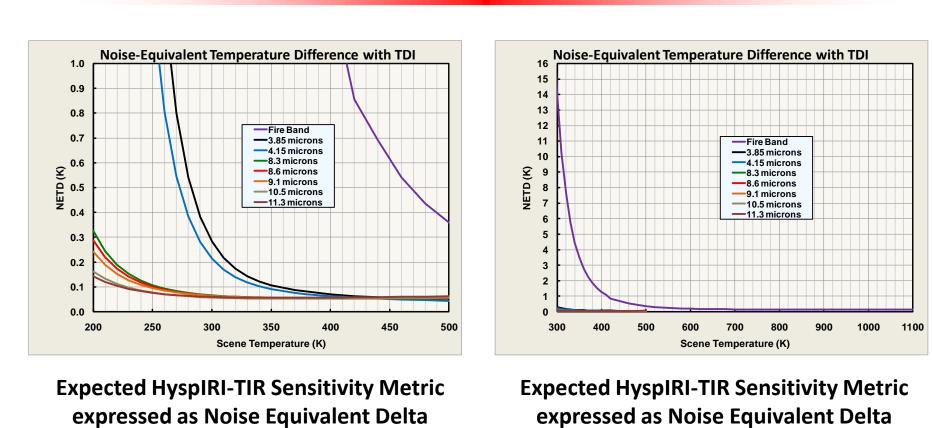
The technology for the HyspIRI-TIR instrument is mature but further work is needed to reduce risk. In particular, the proposed design requires a high sensitivity and high throughput Focal Plane Array (FPA), combined with a scanning mechanism that requires stringent pointing knowledge. The scanning approach, and the high sensitivity and high throughput FPA, are required to meet the revisit time (5 days), the high spatial resolution (60m), and the number of spectral channels (8) specified by the Decadal Survey, and the HyspIRI Science Study Group for the mission. The next step is to reduce the risk associated with the scanning mechanism and the FPA with the development of a laboratory prototype termed the Prototype HyspIRI Thermal Infrared Radiometer (PHyTIR).

PHyTIR will demonstrate that:

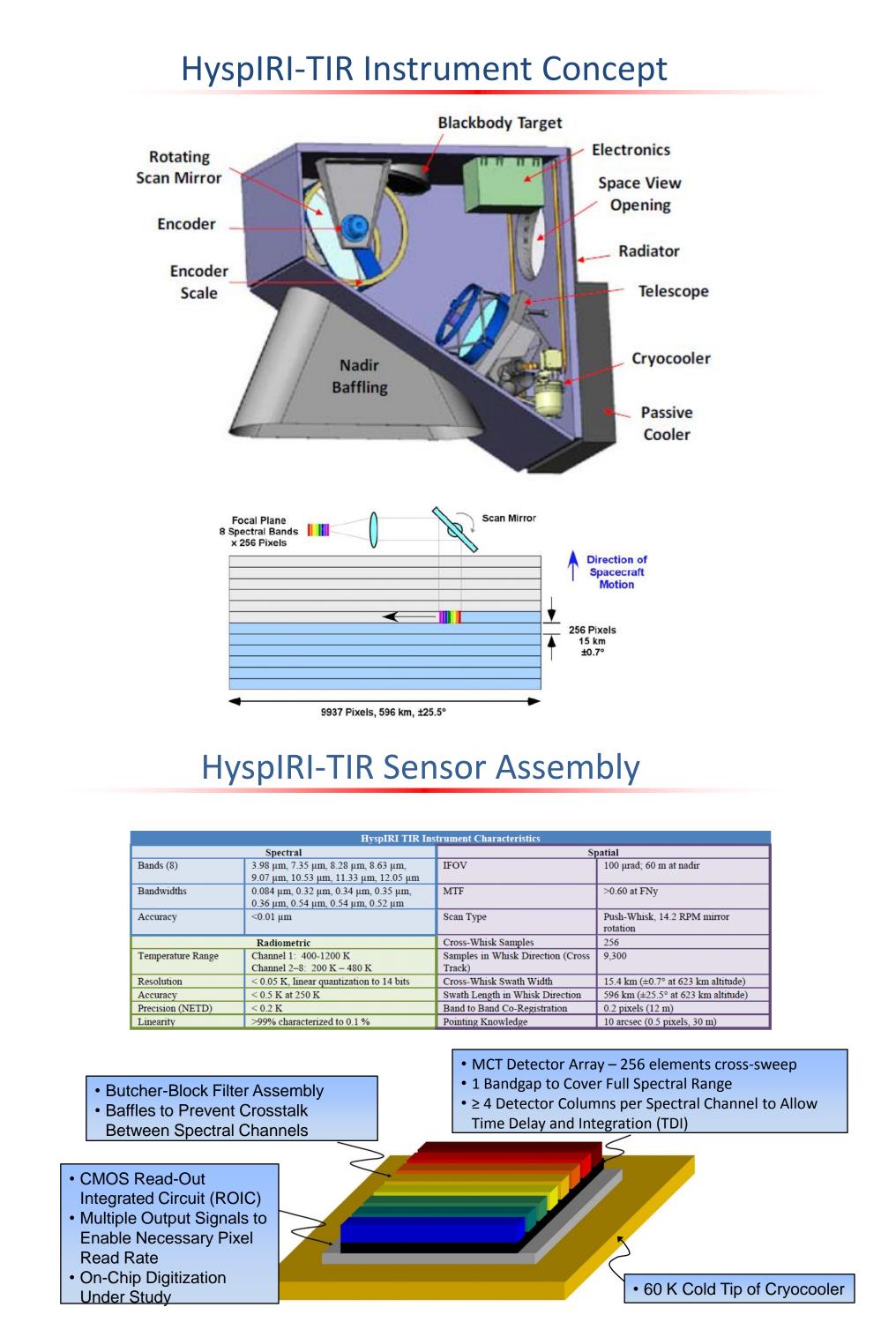
Temperature (NEDT)

- 1.The detectors and readouts meet all signal-to-noise and speed specifications.
- 2.The scan mirror, together with the structural stability, meets the pointing knowledge requirements.
- 3. The long-wavelength channels do not saturate below 480 K.
- 4. The cold shielding allows the use of ambient temperature optics on the HyspIRI-TIR instrument without impacting instrument performance.

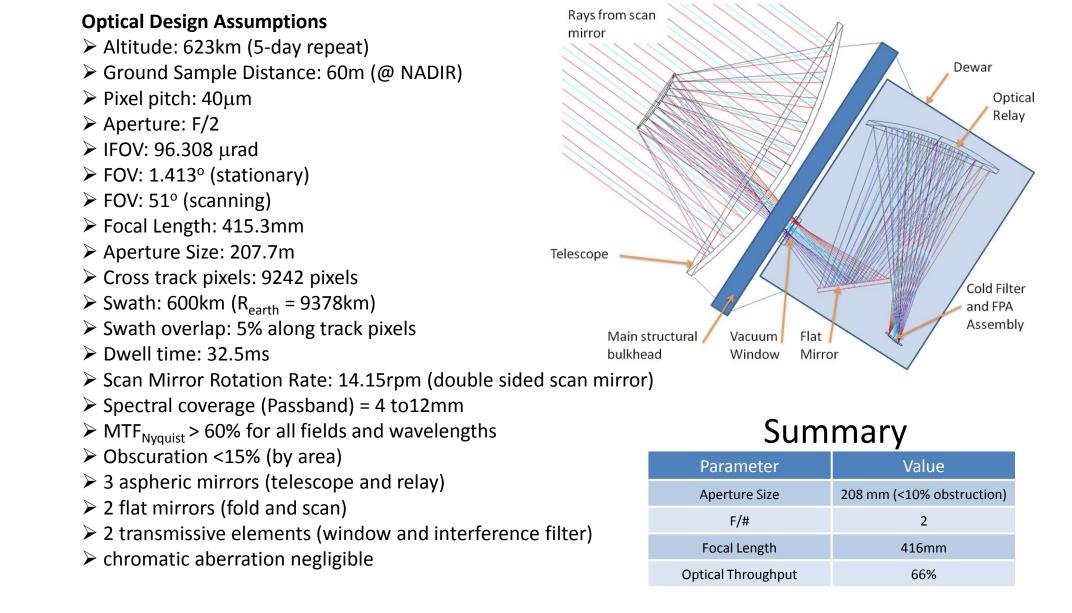
HyspIRI-TIR Sensor Performance

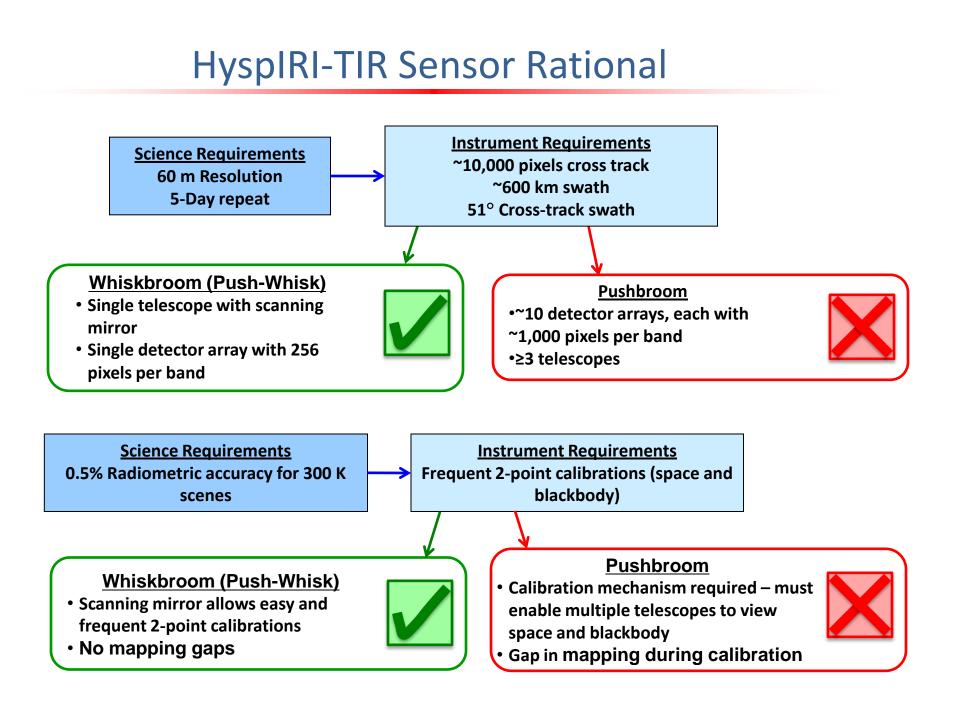


Temperature (NEDT)

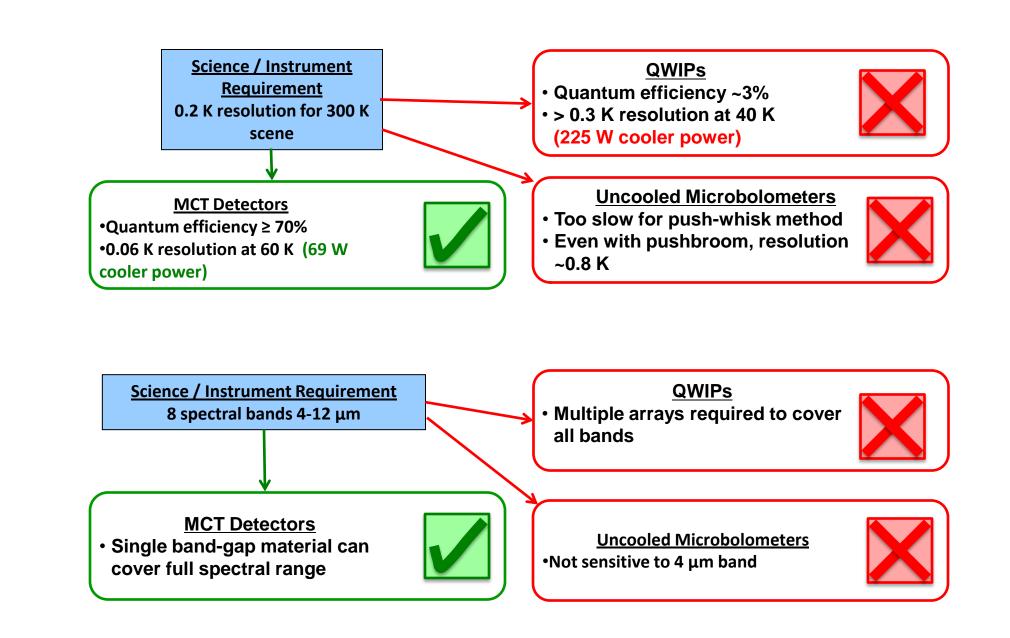


HyspIRI-TIR will use a Butcher-block filter layout on top of focal plane. PHyTIR will not use focal plane filers but will spin a filter wheel in the optical path. **Instrument Optics**

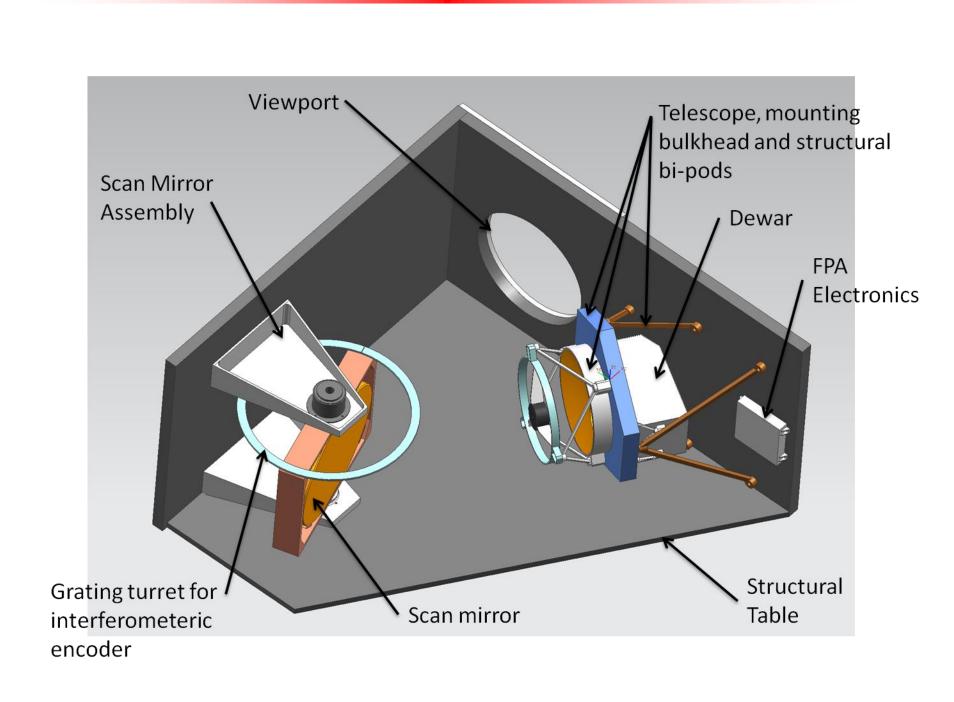


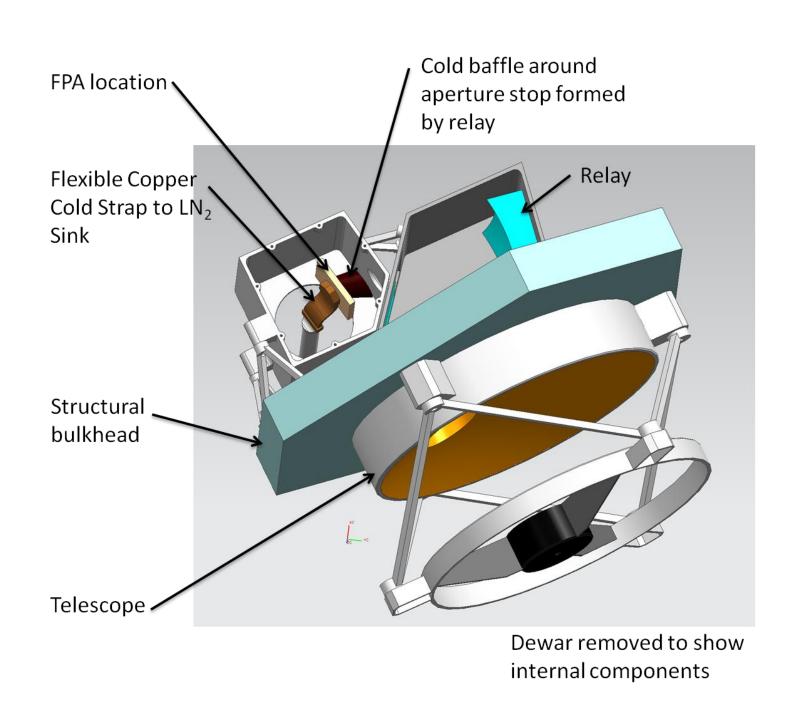


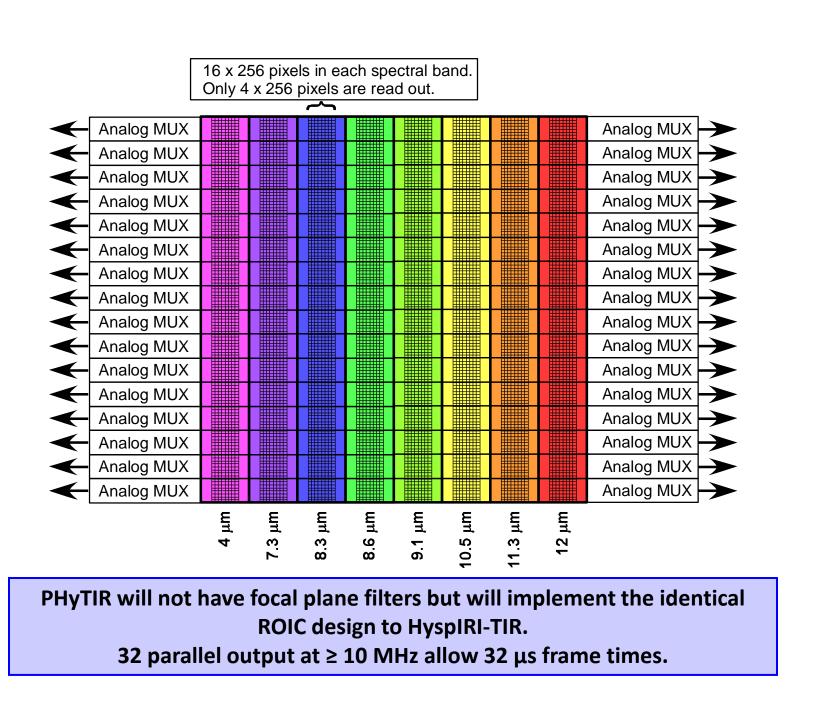
HyspIRI-TIR Sensor Rational



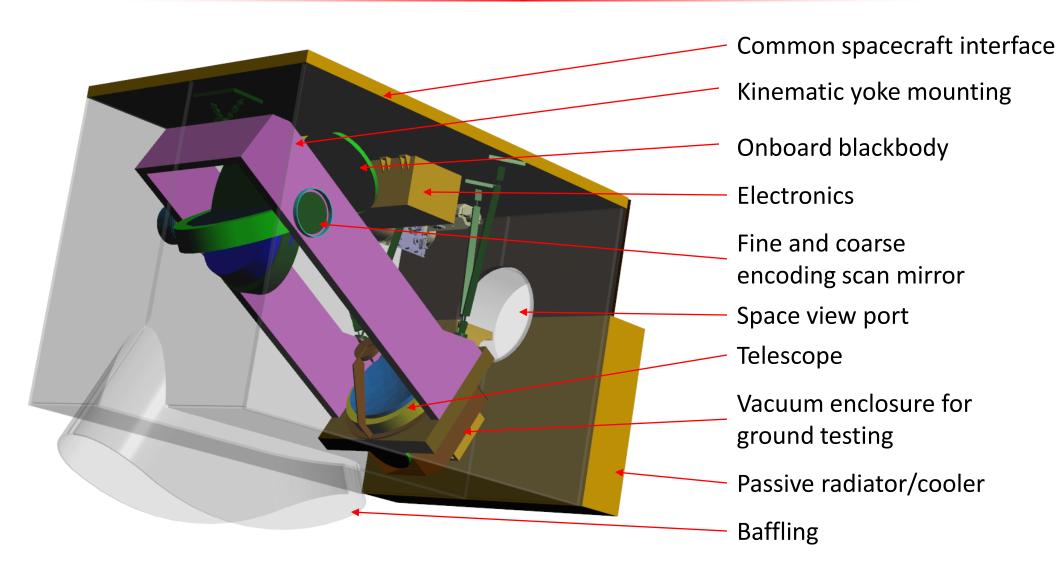
PHyTIR Implementation



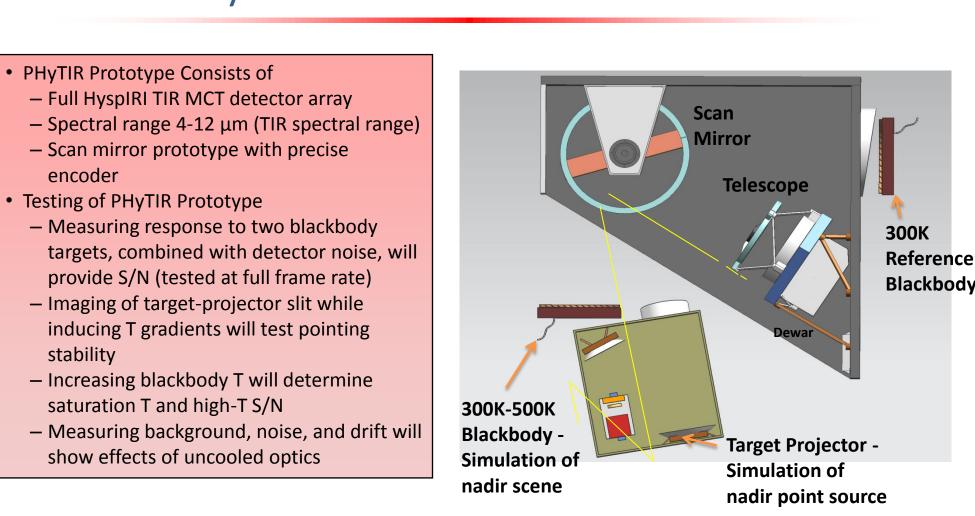




PHyTIR Current Design



PHyTIR Calibration and Validation



PHyTIR Summary

The following steps are currently being undertaken to build PHyTIR:

- 1) Design and Build the Scan Mechanism
- 2) Design and Build a Scan Mirror
- 3) Integrate the Spectral Filters with Focal Plane Array and ROIC
- 4) Assemble the Dewar with external telescope, internal relay and focal plane assembly
- 5) Build the prototype Electronics
- 6) Assemble PHyTIR

Once PHyTIR is assembled it will be used to retire the four key risks as noted earlier. A key part of this effort is the final testing to prove these four key risks.

specifications.

a) Detectors and readout meet all signal-to-noise and speed

- b) Scan mirror and structure meet pointing knowledge requirements.
- c) Long-wavelength channels will not saturate below 480 K.
- d) Background from ambient temperature optics does not affect instrument performance.

This activity will benefit the development of any airborne or spaceborne system that will utilize a high speed scanning mirror coupled with a MCT detector array to obtain a wide swath width, high spatial resolution, thermal infrared measurement with an NE Δ T of approximately 0.2K.

Similar systems have been used in the Moderate Resolution Imaging Spectroradiometer (MODIS), Visible Infrared Imaging Radiometer Suite (VIIRS), Advanced Spaceborne Thermal Emission Radiometer (ASTER) and Landsat (TM5/ETM+) instruments (Barnes et. al. 1998; Mitchel 2008; Ohmae and Kitamure, 1994; Barsi et al. 2003).

However, none of these existing systems has sufficient performance to meet the measurement requirements of the HyspIRI-TIR instrument. PHyTIR will demonstrate that HyspIRI-TIR required high accuracy measurements can be made and help enable both the HyspIRI-TIR instrument as well as other future instruments built by Governments or Commercial Companies that utilize similar technology.

This research was carried out at the Jet Propulsion Laboratory, California Institute of Technology, under a contract with the National Aeronautics and Space Administration. Reference herein to any specific commercial product, process, or service by trade name, trademark, manufacturer, or otherwise, does not constitute or imply its endorsement by the United States Government or the Jet Propulsion Laboratory, California Institute of Technology.